Milestone Review Flysheet

PDR, CDR, FRR

Institution Name	Massachusetts Institute of Technology
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Milestone	FRR

Vehicle Properties	
Diameter (in)	6.28
Length (in)	126.5
Gross Liftoff Weight (lb)	49.27
Launch Lug/button Size	3/16"
Motor Retention	Stainless Steel Rod

Stability Analysis		
Center of Pressure (in from nose)	89.95	
Center of Gravity (in from nose)	77.09	
Static Stability Margin	2.11	
Thrust-to-Weight Ratio	8.3	
Rail Size (in) / Length (in)	1.5" x 1.5", 8ft	

	Recovery System Properties				
	Dr	ogue Parach	ute		
Manufactu	ırer/Model	Sur	plus Military Pilot		
Si	ze		60"		
Altituo	le at Deploym	ent (ft)	5,280		
Velocit	y at Deployme	ent (ft/s)	~20 ft/s depending on wind		
Tern	ninal Velocity	(ft/s)	54.5		
Recov	ery Harness M	[aterial	Tubuluar Nylon		
Harness Size/Thickness (in)		1"			
Recovery Harness Length (ft)		16'			
Harness/Airframe of payload sa		3/8" forged eye bolt in top bot unit. Eye bolt also ebbing that goes to tender			
Kinetic Energy During Descent	Section 1	Section 2	Section 3	Section 4	
(ft-lb)	1615				

Recovery System Properties		
Electronics/Ejection		
Altimeter(s) Make/Model	Featherweight Raven2	
Redundancy Plan	Perfectflite Stratologger, redundant drogue ejection charges. Redundant igniters in Tender Descender on main	
Pad Stay Time (Launch Configuration)	Upwards of 8 hours	

Motor Properties	
Motor Manufacturer	Cesaroni
Motor Designation	L1115
Max/Average Thrust (N/lb)	1713N, 1119N
Total Impulse (N-sec/lb-sec)	5015 Ns
Mass pre/post Burn (lb)	4.4kg, 2.01kg

Ascent Analysis		
Rail Exit Velocity (ft/s)	54	
Max Velocity (ft/s)	594.6	
Max Mach Number	0.53	
Max Acceleration (ft/s^2)	503.3	
Peak Altitude (ft)	5,463	

	Recovery System Properties				
	Main Parachute				
Manufactu	ırer/Model		RocketMan		
Si	ze		16'		
Altitud	de at Deploym	ent (ft)	800'		
Velocit	y at Deployme	ent (ft/s)	54	1.5	
Lan	ding Velocity	(ft/s)	10.98		
Recovery Harness Material		Iaterial	Tubular Nylon		
Harness Size/Thickness (in)		ess (in)	1"		
Recovery Harness Length (ft)		ngth (ft)	3.25'		
Harness/Airframe bay		3/8" Forged eye-nut into top of avionics bay, connected to threaded rod into top of motor			
Kinetic Energy Upon Landing	Section 1	Section 2	Section 3	Section 4	
(ft-lb)	42.65 (body)	.95 (Nose/Sab	3.3 (Quadroto	r)	

Recovery System Properties		
Elec	ctronics/Ejection	
Rocket Locators (Make, Model)	Big Red Bee 70cm Transmitter/2m GPS.	
Transmitting Frequencies	[RC Transmitter]2.4 Ghz (hopping 9 Ch); [3DR]433 Mhz; [Xbee]900 Mhz; [Video Stream]1280 Mhz	
Black Power Mass	3.5	
Drogue Parachute (gram)		
Black Power Mass	0.2 (Tender Descender release	
Main Parachute (gram)	device)	

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	Payload/Science		
Succinct Overview of Payload/Science Experiment	A Custom built composite quadrotor carrying a system of payloads to exhibit object tracking and recognition in dynamic environments and to quantitatively measure high-altitude atmospheric lightning events		
Identify Major Components	The primary payload of the rocket will be the quadrotor which will be equipped with GPS, radio telecommand devices, an IMU, pressure sensors, an onboard computes, CMOS cameras, IR emitters/detectors, and an autonomous flight stabilizer.		
Mass of Payload/Science	The mass of the quadrotor is being estimated to be about 10 pounds.		

Test Plan Schedule/Status		
Ejection Charge Test(s)	12/14/ 2012 (Successful)	
Sub-scale Test Flights	12/15/2012 (Successful)	
Full-scale Test Flights	3/17/2103 (Unsuccessful) Follow up being planned for either 4/6/2013 or 4/13/2013	

Additional Comments		